

16.1: Introduction

We are going to consider the following problem. Two masses, M_1 and M_2 are revolving around their mutual centre of mass C in circular orbits, at a constant distance a apart.

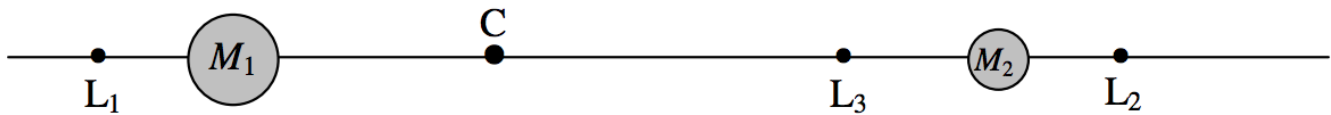


FIGURE IV.4

The orbital period is given by

$$P^2 = \frac{4\pi^2 a^3}{G(M_1 + M_2)} \quad (16.1.1)$$

and the angular orbital speed is given by

$$\omega^2 = \frac{G(M_1 + M_2)}{a^3}. \quad (16.1.2)$$

I establish the following notation.

Mass ratio:

$$\frac{M_1}{M_2} = q. \quad (16.1.3)$$

Mass fraction:

$$\frac{M_1}{M_1 + M_2} = \mu. \quad (16.1.4)$$

They are related by

$$q = \frac{\mu}{1 - \mu} \quad (16.1.5)$$

and

$$\mu = \frac{q}{1 + q}. \quad (16.1.6)$$

We note the following distances:

$$M_1 C = (1 - \mu)a, \quad M_2 C = \mu a. \quad (16.1.7)$$

We ask ourselves the following question: Are there any points on the line passing through the two masses where a third body of negligible mass could orbit around C with the same period as the other two masses; i.e. it would remain on the line joining the two main masses?

In fact there are three such points, and they are known as the *collinear lagrangian points*. (The collinear points were discussed by Euler before Lagrange, but Lagrange took the problem further and discovered an additional two points not collinear with the masses, and the five points today are generally all known as the lagrangian points. We shall discuss the additional points in Section 16.2.) I have marked the three points in figure XVI.4 with the letters L_1 , L_2 and L_3 .

Nomenclature

There are evidently $3! = 6$ ways in which I could choose the subscripts. Often today, the inner lagrangian point is labelled L_1 and the outer points are labelled L_2 and L_3 . This seems to me to lack logic, and I choose to label the inner point L_3 , and the outer points associated with M_1 and M_2 are then L_1 and L_2 respectively. Incidentally, I am not making any assumption about which of the two main bodies is the more massive.

Let us deal first with L_1 . Let us suppose that the distance from C to L_1 is xa .

A particle of mass m at L_1 is subject (in a co-rotating reference frame) to three forces, namely the gravitational attractions from the two main bodies, and the centrifugal force acting away from C. If this body is to be in equilibrium, we must have

$$\frac{GM_1m}{[(x-1+\mu)a]^2} + \frac{GM_2m}{[(x+\mu)a]^2} = mxa\omega^2. \quad (16.1.8)$$

On making use of Equations 16.1.2 and 16.1.4, we find that this Equation becomes

$$\frac{\mu}{(x-1+\mu)^2} + \frac{1-\mu}{(x+\mu)^2} = x. \quad (16.1.9)$$

After manipulation, this becomes

$$a_0 + a_1x + a_2x^2 + a_3x^3 + a_4x^4 + x^5 = 0, \quad (16.1.10)$$

where

$$a_0 = -1 + 3\mu - 3\mu^2, \quad (16.1.11)$$

$$a_1 = 2 - 4\mu + \mu^2 - 2\mu^3 + \mu^4, \quad (16.1.12)$$

$$a_2 = -1 + 2\mu - 6\mu^2 + 4\mu^3, \quad (16.1.13)$$

$$a_3 = 1 - 6\mu + 6\mu^2 \quad (16.1.14)$$

and

$$a_4 = -2 + 4\mu. \quad (16.1.15)$$

Although Equation 16.1.10 is a quintic Equation, it has just one real root for positive μ .

The positions of L_2 and L_3 can be found by exactly similar arguments – you just have to take care with the directions and distances of the two gravitational forces.

For L_2 , the coefficients are the same as for L_1 , except

$$a_1 = -2 + 4\mu + \mu^2 - 2\mu^3 + \mu^4, \quad (16.1.6)$$

$$a_2 = -1 - 2\mu + 6\mu^2 - 4\mu^3 \quad (16.1.17)$$

and

$$a_4 = 2 - 4\mu. \quad (16.1.18)$$

For L_3 the coefficients are

$$a_0 = 1 - 3\mu + 3\mu^2 - 2\mu^3, \quad (16.1.1)$$

$$a_1 = 2 - 4\mu + 5\mu^2 - 2\mu^3 + \mu^4, \quad (16.1.2)$$

$$a_2 = 1 - 4\mu + 6\mu^2 - 4\mu^3, \quad (16.1.3)$$

$$a_3 = 1 - 6\mu + 6\mu^2 \quad (16.1.4)$$

$$a_4 = 2 - 4\mu. \quad (16.1.5)$$

(Reminder: When computing any of these polynomials, write them in terms of nested parentheses. See Chapter 1, Section 1.5.)

It is also of interest to see the equivalent potential (gravitational plus centrifugal). The expression for gravitational potential energy is, as usual, $-GMm/r$, where r is the distance from the mass M . The expression for the centrifugal potential energy is $-\frac{1}{2}m\omega^2r^2$, where r is the distance from the centre of mass. The negative of the derivative of this expression is $m\omega^2r$ which is the usual expression for the centrifugal force. When we apply these principles to the system of two masses under consideration, we obtain the following expression for the equivalent potential (which, in this section, I'll just call V rather than V').

$$V = -\frac{GM_1}{|x+1-\mu|a} - \frac{GM_2}{|x-\mu|a} - \frac{1}{2}x^2a^2\omega^2. \quad (16.1.24)$$

On making use of Equations 16.1.2 and 16.1.4, we find that this Equation becomes

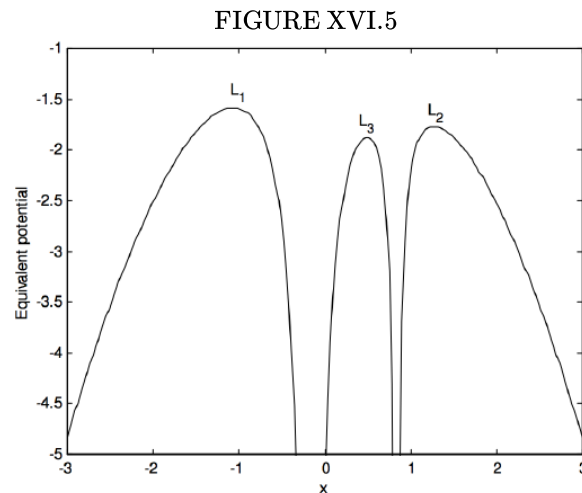
$$W = -\frac{\mu}{|x+1-\mu|} - \frac{1-\mu}{|x-\mu|} - \frac{x^2}{2}, \quad (16.1.25)$$

where

$$W = V \div \left(\frac{G(M_1 + M_2)}{a} \right). \quad (16.1.26)$$

Setting the derivatives of this expression to zero gives, of course, the positions of the lagrangian points, for these are equilibrium points where the derivative of the potential is zero. Figure XVI.5 shows the potential for a mass ratio $q = 5$. Note that, in the line joining the two masses, the equivalent potential at the lagrangian points is a maximum, and therefore these points, while equilibrium points, are unstable. We shall see in Section 16.6 that the points are actually saddle points. While several spacecraft are in orbit or are planned to be in orbit around the collinear lagrangian points (e.g. SOHO at the interior lagrangian point, and MAP at L_2), continued small expenditure of fuel is presumably needed to keep them there.

It will be of interest to see how the positions of the lagrangian points vary with mass fraction. Indeed mass can be transferred from one member of a binary star system to the other during the evolution of a binary star system. We shall discuss a little later how this can happen. For the time being, without worrying about the exact mechanism, we'll just vary the mass fraction and see how the positions of the lagrangian points vary as we do so. However, if mass is transferred from one member of a binary star system to the other,



and if there are no external torques on the system, the angular momentum L of the system will be conserved, and, to ensure this, the separation a of the two stars changes with mass fraction.

✓ Example 16.1.1

Show that, for a given orbital angular momentum L of the system, the separation a of the components varies with mass fraction according to

$$a = \frac{L^2}{GM^3\mu^2(1-\mu)^2}. \quad (16.1.27)$$

Solution

Here $M = M_1 + M_2$ is the total mass of the system. In figure XVI.6 I have used this Equation, plus Equations 16.1.10 and 16.1.7, to compute the distances of M_2 , C, and the three lagrangian points from M_1 as a function of mass fraction. The unit of distance in figure XVI.6 is $16L^2/(GM^3)$, which is the separation of the two masses when the two masses are equal. Each of

moves from M_1 towards M_2 , it is subject to a *Coriolis force* (see section 4.9 of Classical Mechanics), which sends it around M_2 in an *accretion disc*. During this process the total angular momentum of the system is conserved (provided no mass is lost from the system) but this must now be shared between the orbital angular momentum of the two stars and the angular momentum of the accretion disc. However, as long as the latter is a relatively small contribution to the total angular momentum, conservation of orbital angular momentum remains a realistic approximation.

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